Blade Boundary Layer Resolved Computations of the NREL 5MW Rotor in a Realistic Atmospheric Boundary Layer using Hybrid URANS-LES

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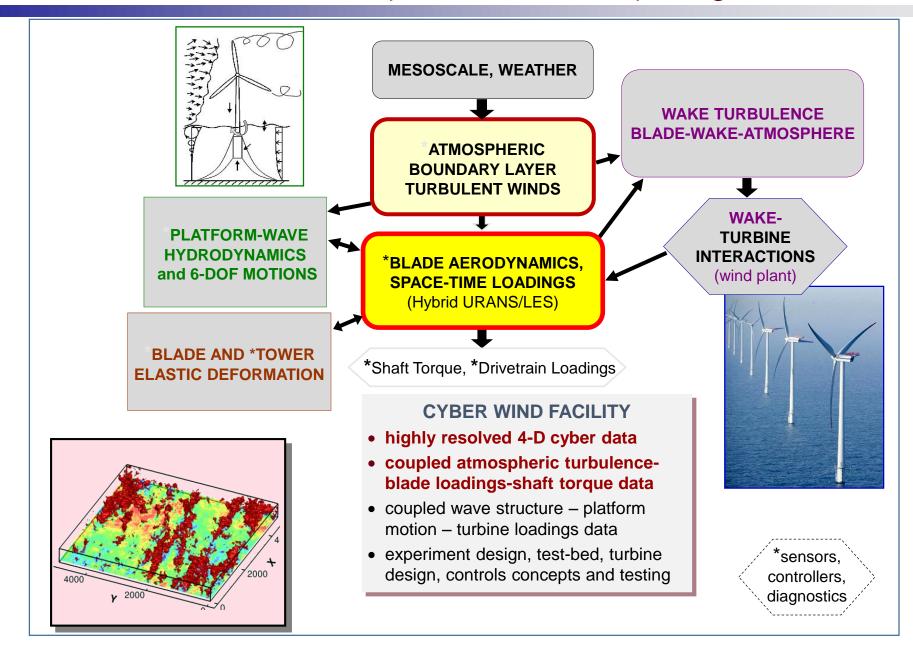
2nd Symposium on OpenFOAM in Wind Energy 19 May 2014, Boulder, CO

*Supported by:

- DOE (EERE) Offshore Wind Technology Development Program
- National Science Foundation Energy for Sustainable Program
- Penn State Applied Research Laboratory
- Penn State: Institutes of Energy and the Environment, CyberScience Institute, Aerospace and Mechanical Engineering
- Computer Resources: NSF XSEDE program; DOE Oak Ridge National Laboratory

Penn State Cyber Wind Facility Program





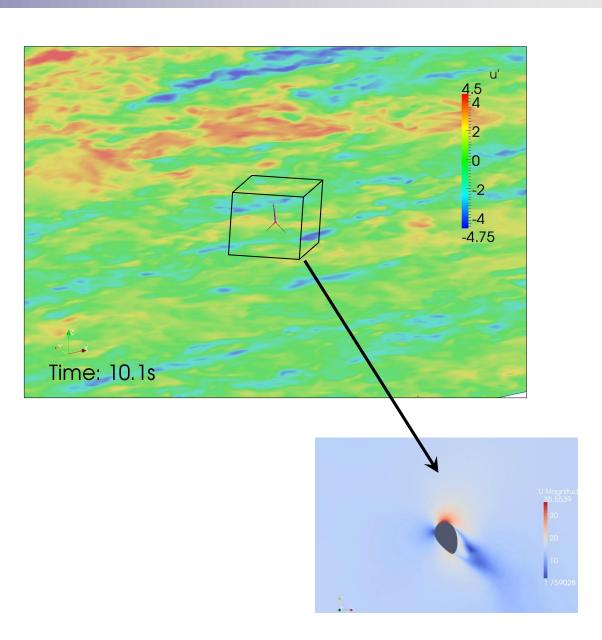
Motivation and objective



Wind turbines fail sooner than expected

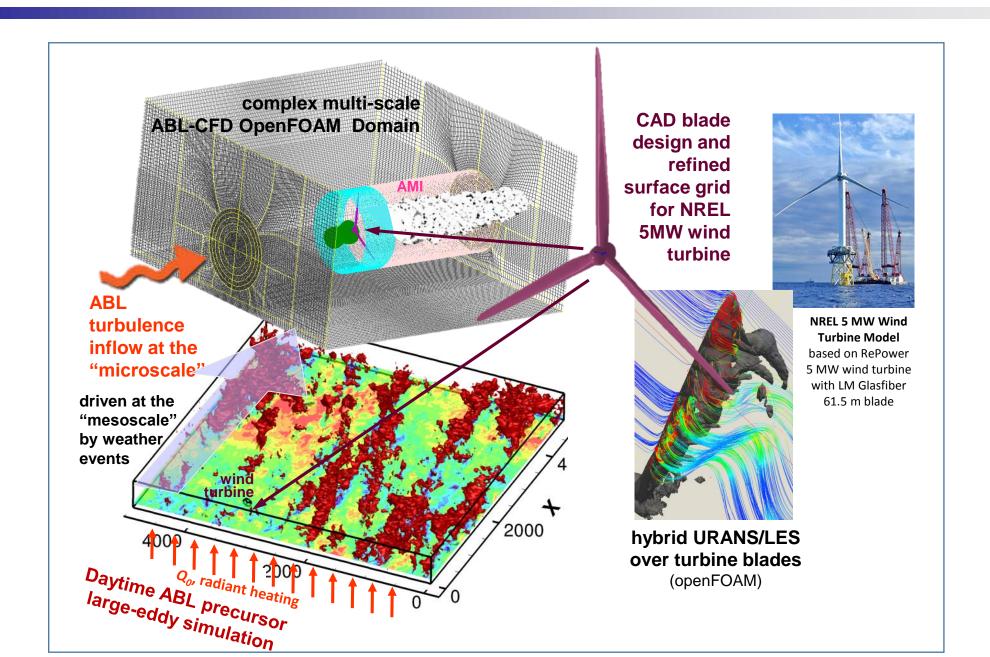
- Gearboxes, blades, shafts, ...
 Atmospheric Boundary Layer (ABL)
- Size of most energetic turbulence structures ~ wind turbine disk

Study the response of blade boundary layer to forcing by Atmospheric Turbulence



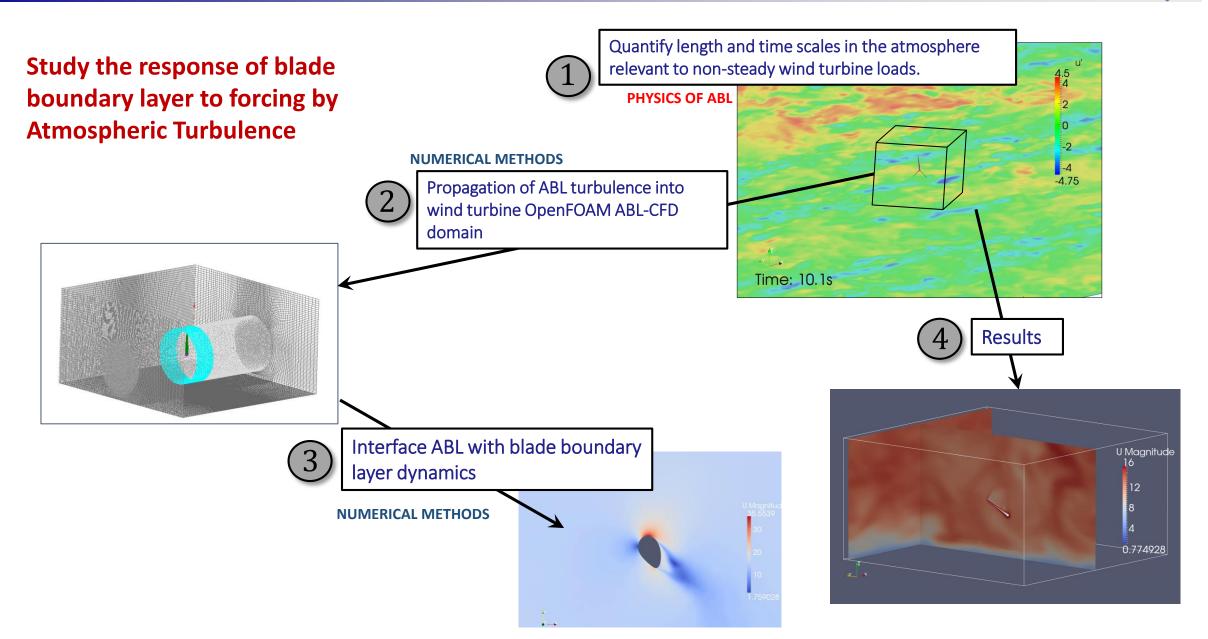
The Current Cyber Wind Facility





Outline

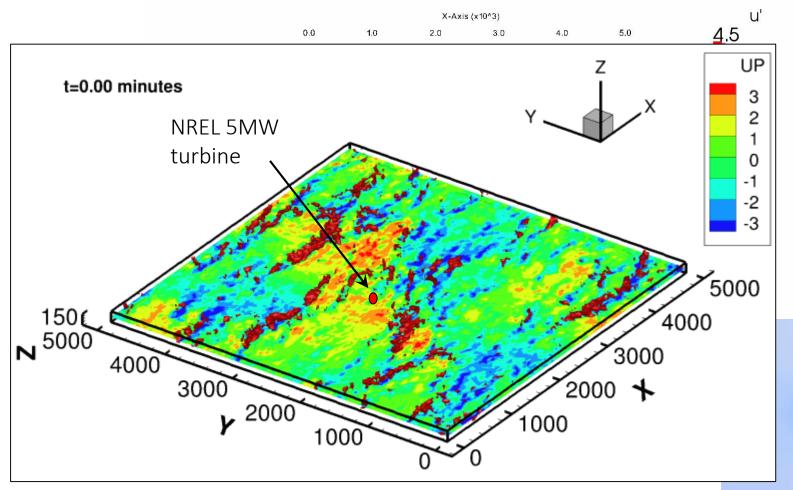


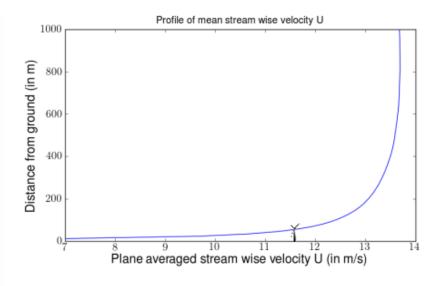


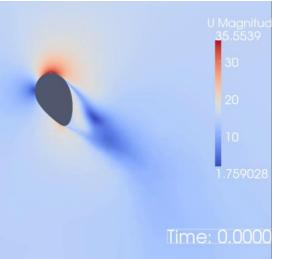
Turbulence structure in the Atmospheric Surface Layer



Moderately Convective Boundary Layer

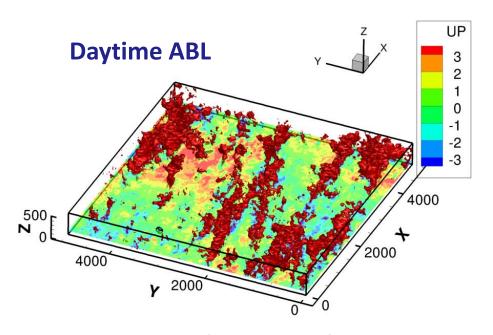






Atmospheric Boundary Layer - LES





Mesoscale – weather $\sim 10 - 100 \ km$

ABL - Eddies $\sim 100 m$

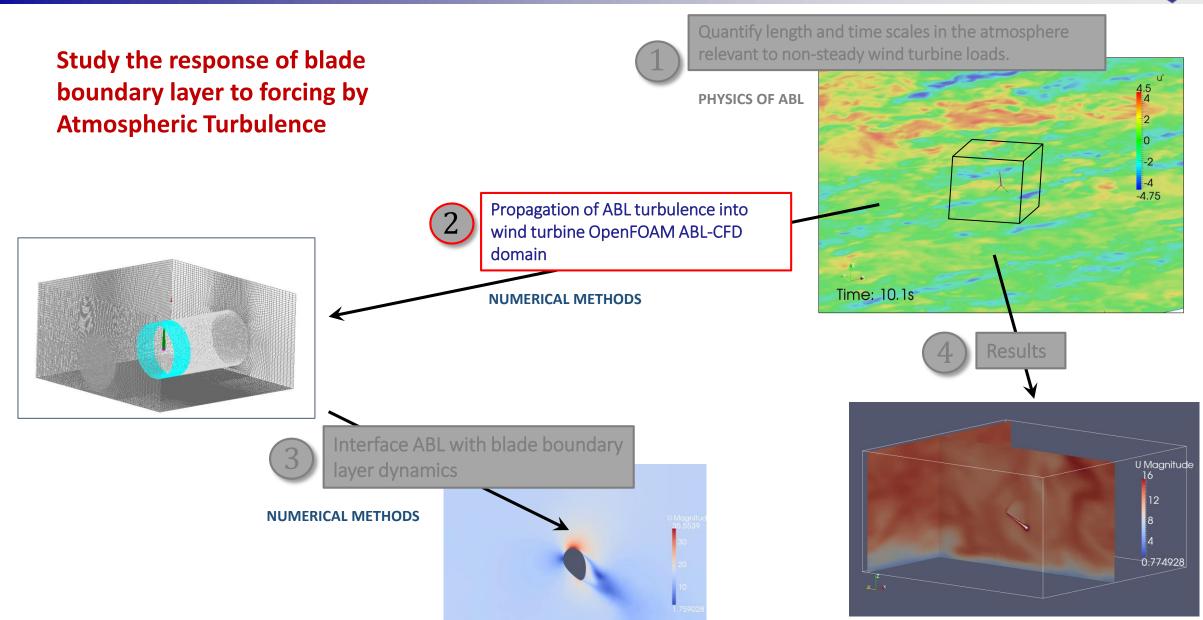
ABL - Code

- Pseudo-spectral low dissipation
- Highly parallel –
 Sullivan and Patton (2008)
- Domain 5km x 5km x 2km
- Grid 512 x 512 x 256
- 14 m/s Mean velocity @ Hub Height– Region III

Blade boundary layer $\sim 1 mm$

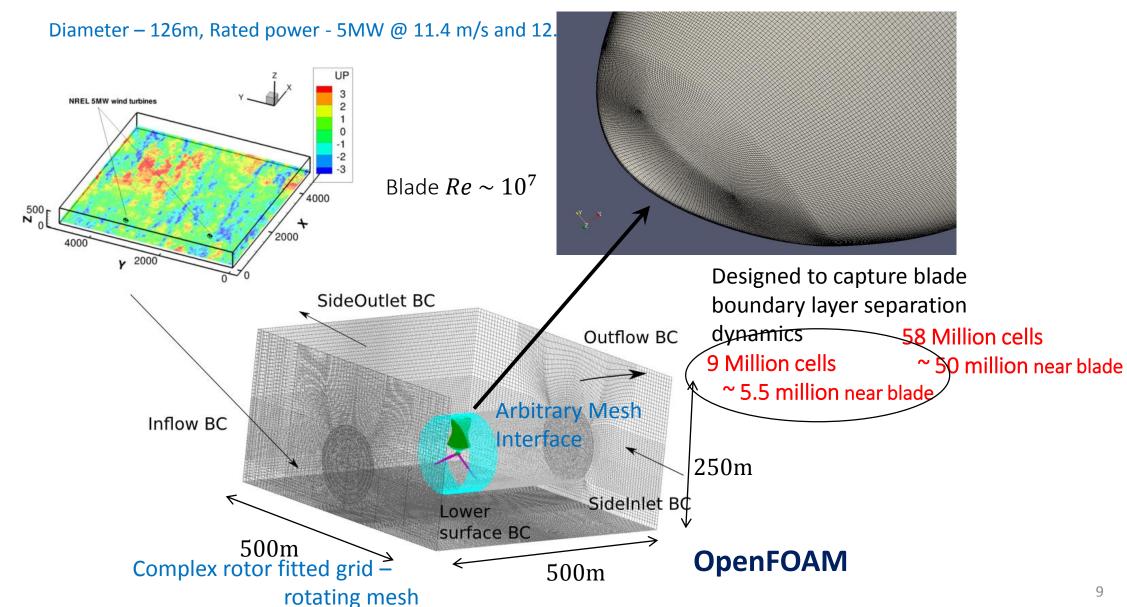
Outline





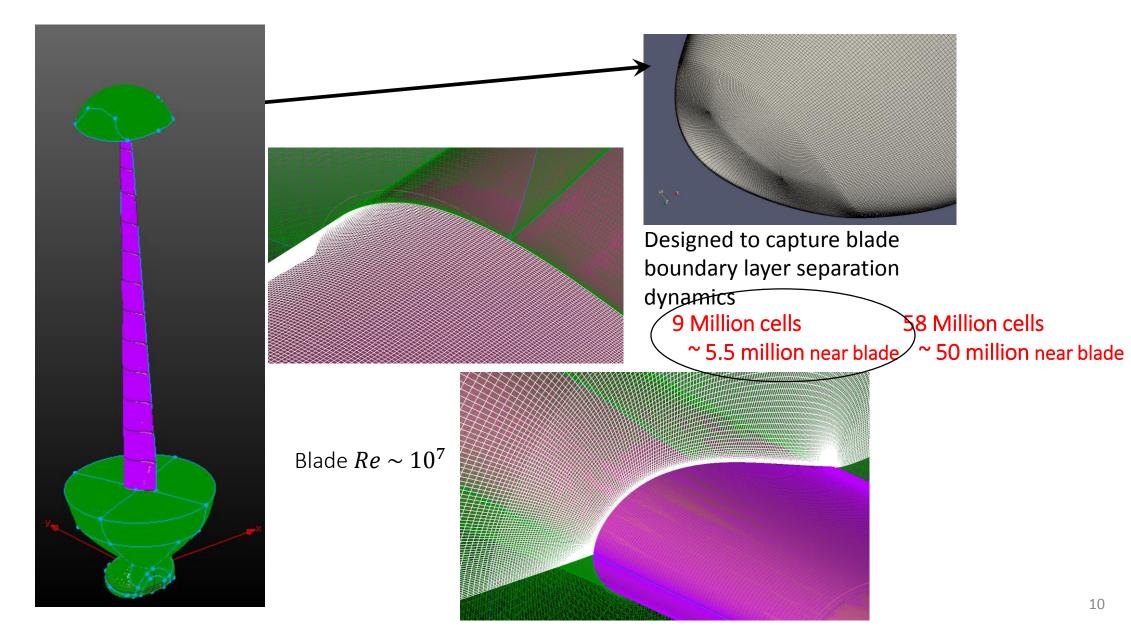
Design of Wind Turbine Geometry and Grid





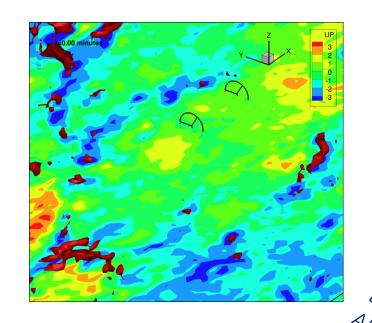
Design of Wind Turbine Geometry and Grid





Coupling LES of ABL to CFD around wind turbine





2010-2011

Collaboration with Dr. Churchfield and Dr. Moriarty

Coriolis *Momentum:* $\frac{\partial U_i}{\partial t} + \frac{\partial U_i U_j}{\partial x_i} = -\frac{1}{\rho_0} \langle \frac{\partial P}{\partial x_i} \rangle - \frac{1}{\rho_0} \frac{\partial P^*}{\partial x_i} - 2\Omega_i U_j \epsilon_{ijk}$ $(\Theta - \langle \Theta \rangle)$ **Buoyancy** -

Boussinesq

Designed to capture blade ABL algorithm to finite boundary layer separation dynamics

SideOutlet BC

OpenFOAM (ABL) Outflow BC Arbitrary Mesh Interface 250m

Blade $Re \sim 10^7$

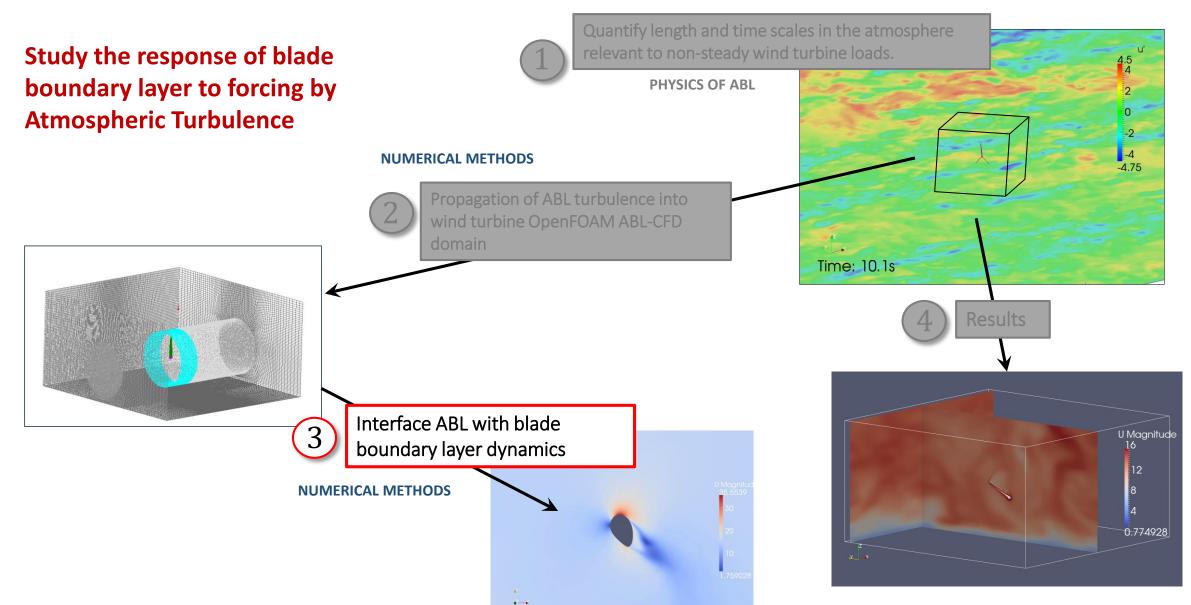
Complex rotor fitted grid – rotating mesh

Inflow BC

SideInlet BC Lower surface BC

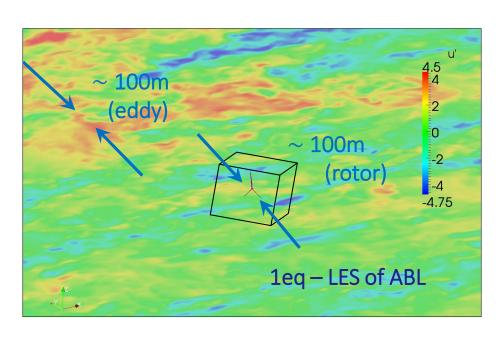
Outline

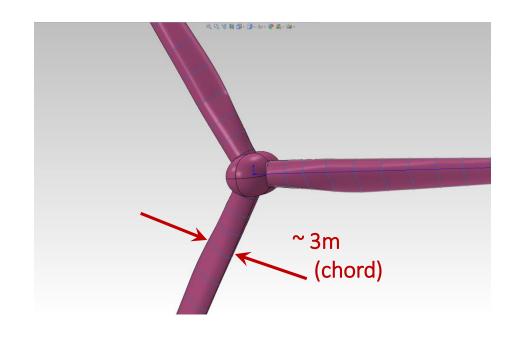




Modeling range of length scales - ABL to Blade boundary layer







Mesoscale – weather $\sim 10-100km$

Rotor - ABL Turbulence $\sim 100m$

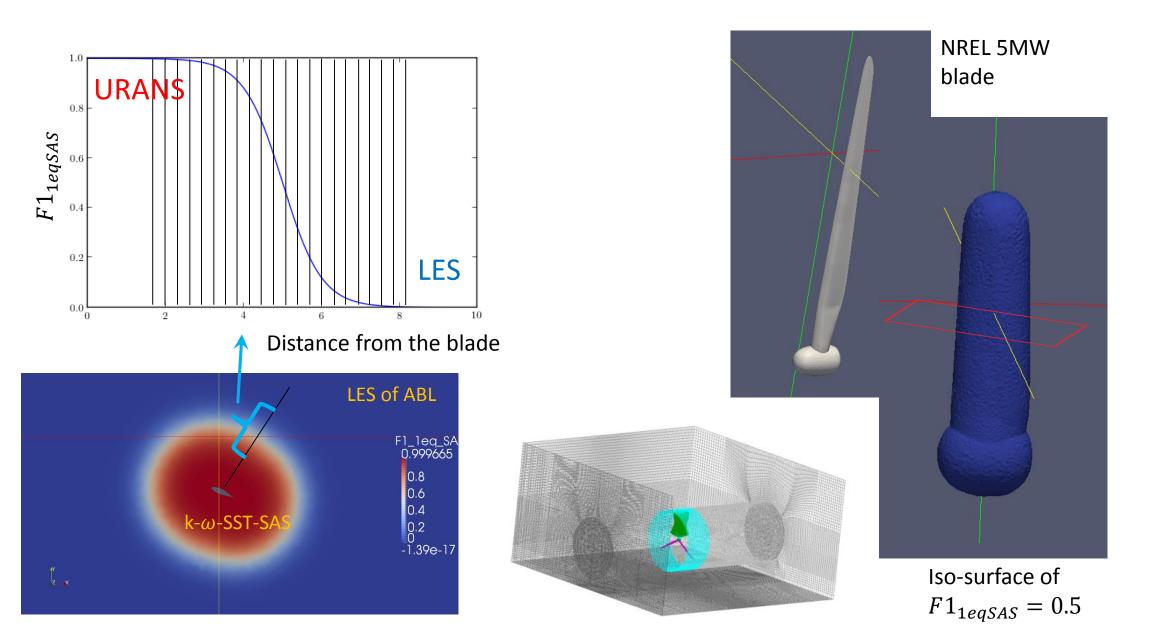
Blade chord $\sim 1m$

Blade boundary layer $\sim 1mm$

(boundary layer) $\sim 10^{-3}$ m $\sim 10^{-3}$ m

New method to blend LES of ABL with Hybrid URANS/LES near blade

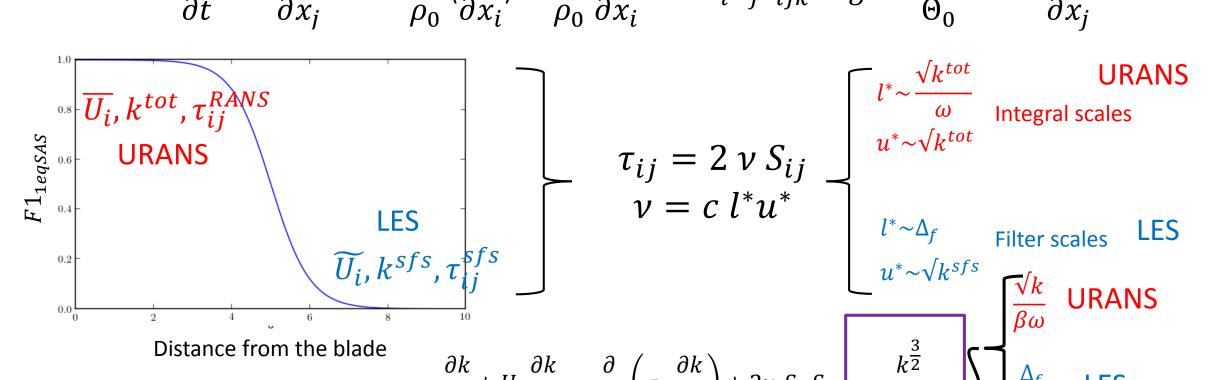




New method to blend LES of ABL with Hybrid URANS/LES near blade



$$Momentum: \frac{\partial U_i}{\partial t} + \frac{\partial U_i U_j}{\partial x_j} = -\frac{1}{\rho_0} \langle \frac{\partial P}{\partial x_i} \rangle - \frac{1}{\rho_0} \frac{\partial P^*}{\partial x_i} - 2\Omega_i U_j \epsilon_{ijk} - g \frac{(\Theta - \langle \Theta \rangle)}{\Theta_0} - \frac{\partial \tau_{ij}}{\partial x_j}$$



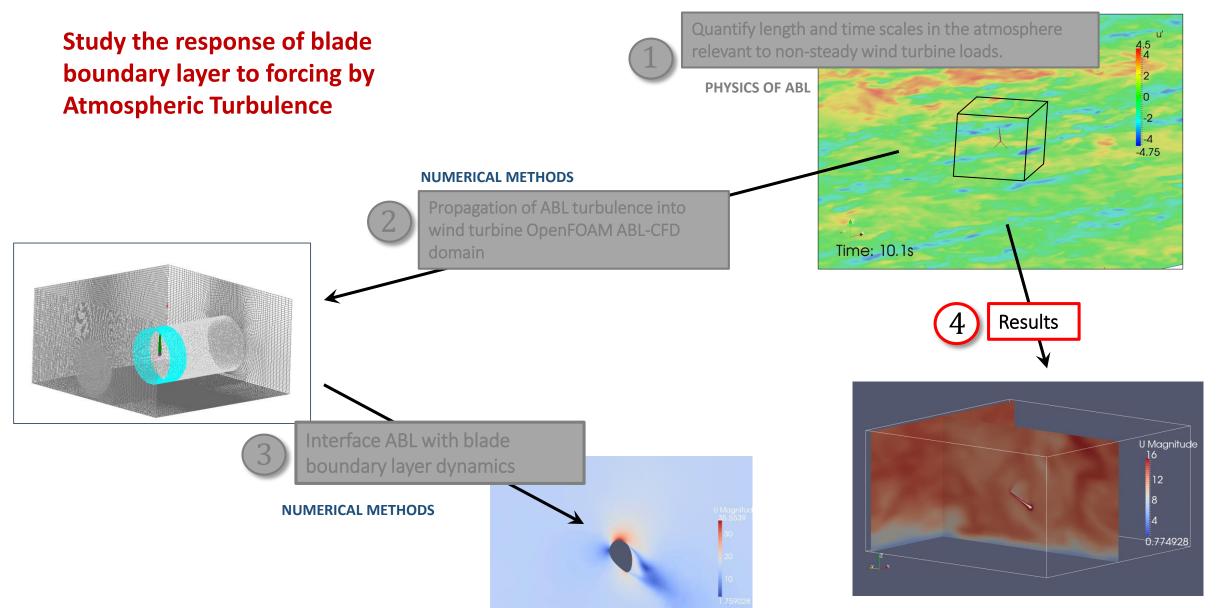
OpenFOAM turbulence model class $\frac{\partial k}{\partial t} + U_j \frac{\partial k}{\partial x_j} = \frac{\partial}{\partial x_j} \left(\sigma_k \frac{\partial k}{\partial x_j} \right) + 2\nu_t S_{ij} S_{ij}$

Advection

Diffusion **Production** **Dissipation term**

Vijayakumar et. al. AIAA 2014-0867

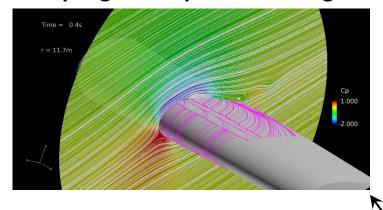




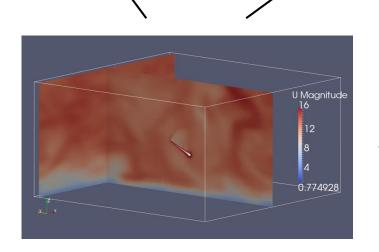
OpenFOAM tools for data analysis



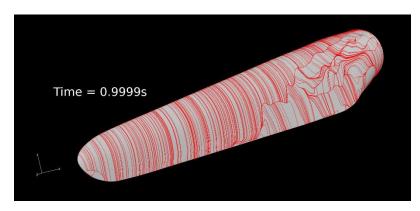
Sampling on cut planes rotating with the inner domain



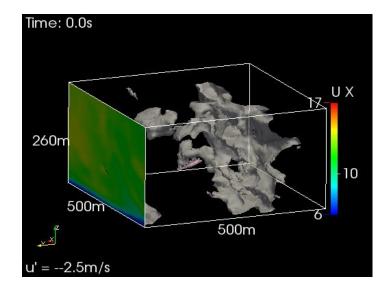




Blade surface data without interpolation

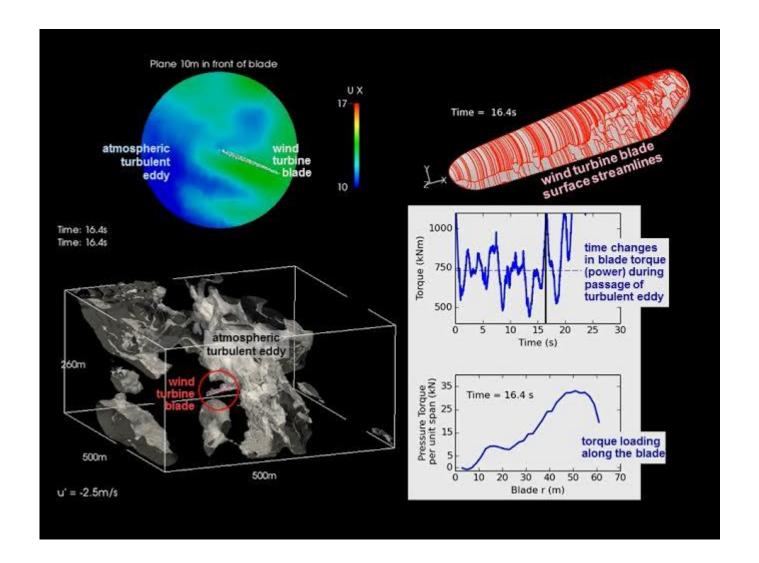


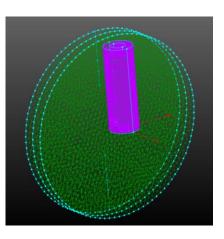
Interpolation to a coarse mesh for volumetric analysis



The largest fluctuations on the wind turbine loads are due to ABL structures ~ 0(rotor disk).



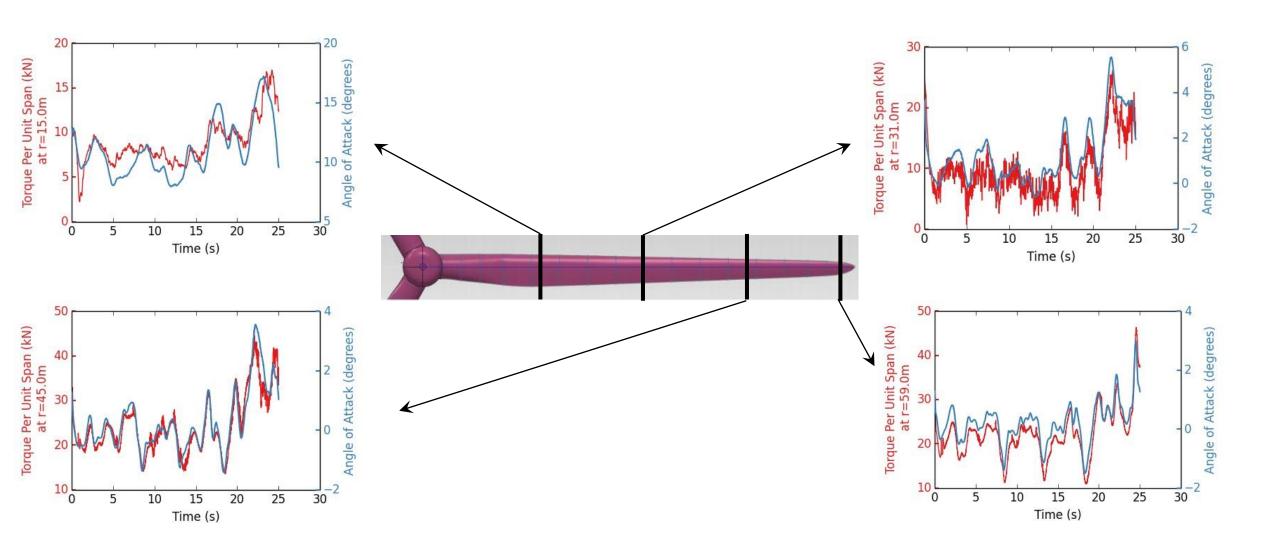




Disk 10m in front of blade

Atmospheric turbulence causes fluctuations in the integrated loads primarily through changes in the angle of attack.

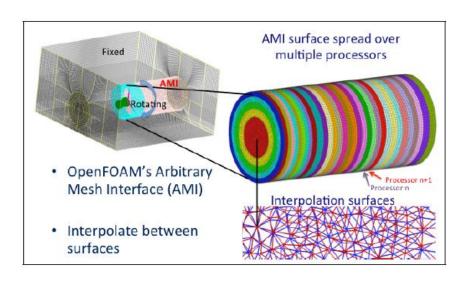




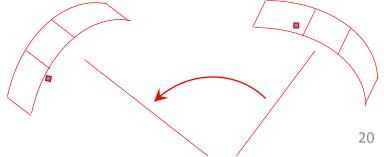
Particular problems with OpenFOAM implementation



 Scaling to large number of cores with AMI – Adam Lavely



- File I/O at large # cores Too many files Working on a HDF5 based solution with Dr. Anirban Jana and Si Liu @ XSEDE
- \times Interpolation near the blade surface grid sizes $\sim 1~\mu m$



Conclusions



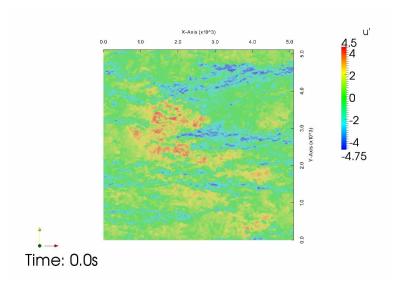
Energy containing scales in ABL

- 1
- ~ NREL 5MW turbine disk
- ~ Multiple rotation time scales

2

Simulation across scales

 \sim 9 orders of magnitude.



The turbulence structures $\sim O(rotor\ disk)$ cause the largest fluctuations in the integrated loads primarily through changes in the angle of attack.

Acknowledgements



- "Kulakowski Travel Award" Mech. Engg. @ PSU
- Dr. Eric Paterson Virginia Tech
- Fieldview University Partners Program
 - Dr. Earl Duque
- NSF XSEDE Dr. Anirban Jana @ PSC